In the Abstract

Please delete the Abstract as presented in the underlying International Application No. PCT/DE2004/000808, and replace it with the following new Abstract:

ABSTRACT

The invention relates to an inlet An abradable coating for gas turbines is provided. Said inlet coating acts as a seal for The abradable coating is used for sealing a radial gap between a casing housing (11) of the gas turbine and rotating blades (10) of the latter same. , the The abradable coating (13) is being applied to the housingcasing. According to the invention, the inlet coating (13) consists of at least one layer, The abradable coating (13) is at least single-layered, at least one outer layer of the inlet abradable coating (13) being produced from a material with having a magnetoplumbite structure, preferably lanthanum hexaaluminate.